



NATIONAL ENERGY TECHNOLOGY LABORATORY



Overview of NETL's High Temperature Heat Transfer and Film Cooling Test Facility

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National Energy Technology Laboratory

¹University of Pittsburgh

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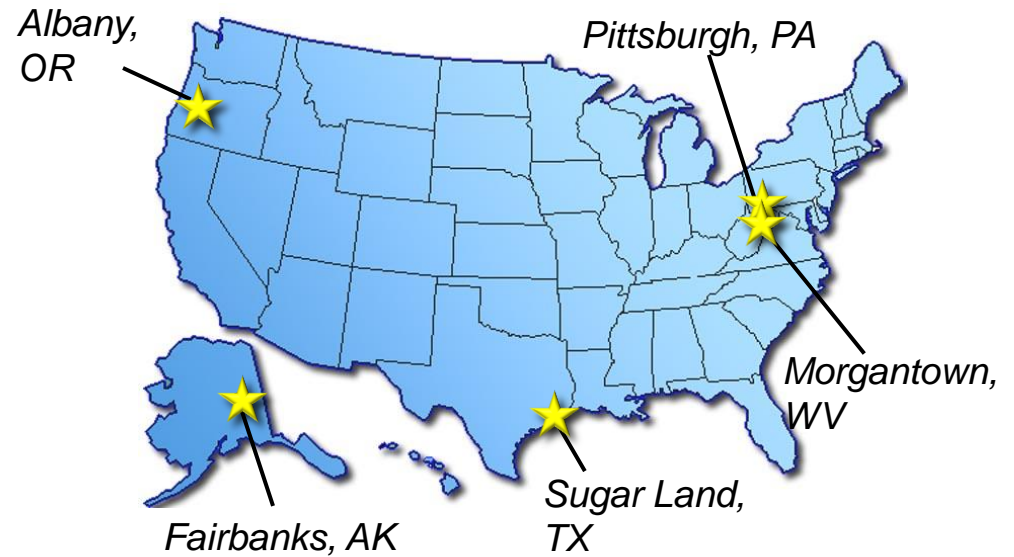
Columbus, OH



National Energy Technology Laboratory

MISSION

*Advancing energy options
to fuel our economy,
strengthen our security, and
improve our environment*



Oregon



Pennsylvania



West Virginia

NETL Full-Spectrum Capacity for Innovation

1

DOE national laboratory dedicated to energy RD&D

- *Economy-wide perspective on energy challenges*
- *100 years of energy technology RD&D experience*

2

Only government owned & operated DOE national laboratory

- *Unbiased approach reflecting national priorities*
- *Direct taxpayer accountability for funding*

3

Combination of research expertise and contracting capabilities

- *Expertise & equipment to perform & evaluate RD&D*
- *Collaboration, contracting, counsel, communication*

4

Established relationships within academia, industry & governments

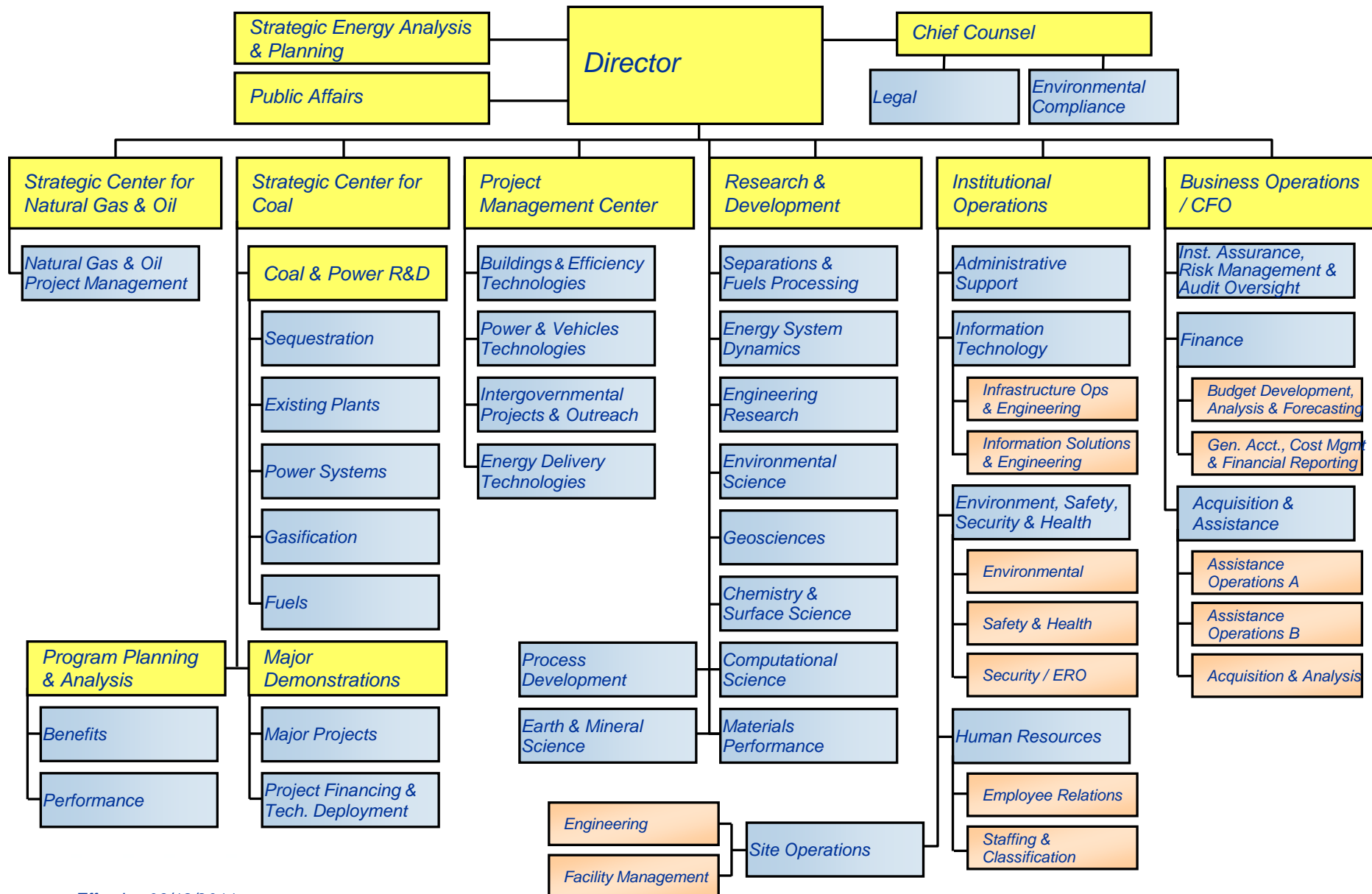
- *Broad vision of the energy technology landscape*
- *Roundhouse for intellectual capital & tech transfer*

5

Market-facing basic & applied science

- *Explicit targets to deliver technology innovations*
- *Public/private cost sharing*

National Energy Technology Laboratory



Regional University Alliance

NETL's Institute for Advanced Energy Studies



*Carnegie
Mellon
University*



*West Virginia
University*

*Virginia
Polytechnic
Institute*



*Pennsylvania
State
University*



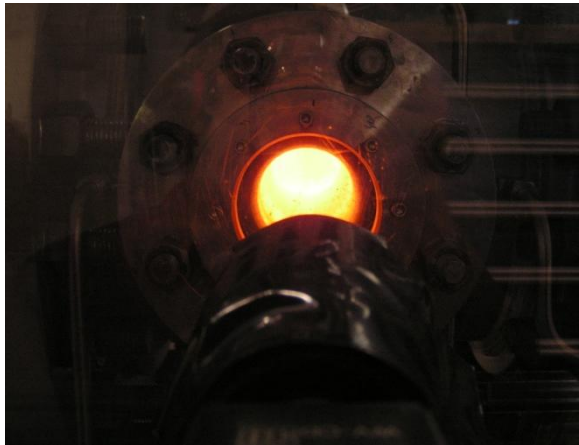
*University of
Pittsburgh*



*Leveraging National Lab and University-Based Scientific and Engineering
Assets to Address Significant National Energy Issues*

Outline

- **Overview high temperature/high pressure test facility**
 - Hardware and facility capabilities
 - Experimental approach
 - Optical surface temperature mapping
 - Lessons learned
- **Overview results of collaboration with University of Pittsburgh**
- **Future work**
- **Summary**

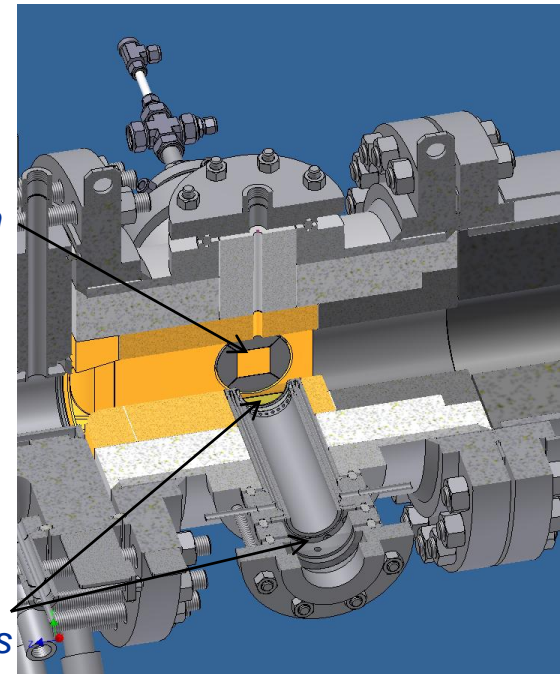


Test Coupon

Hot gas flow

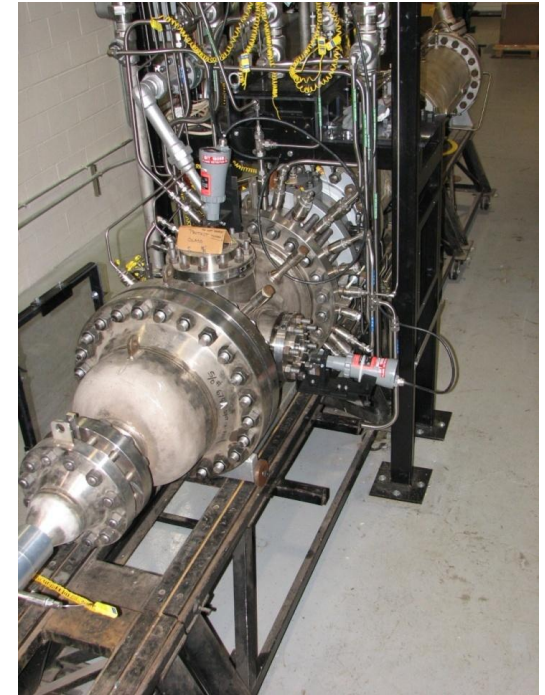
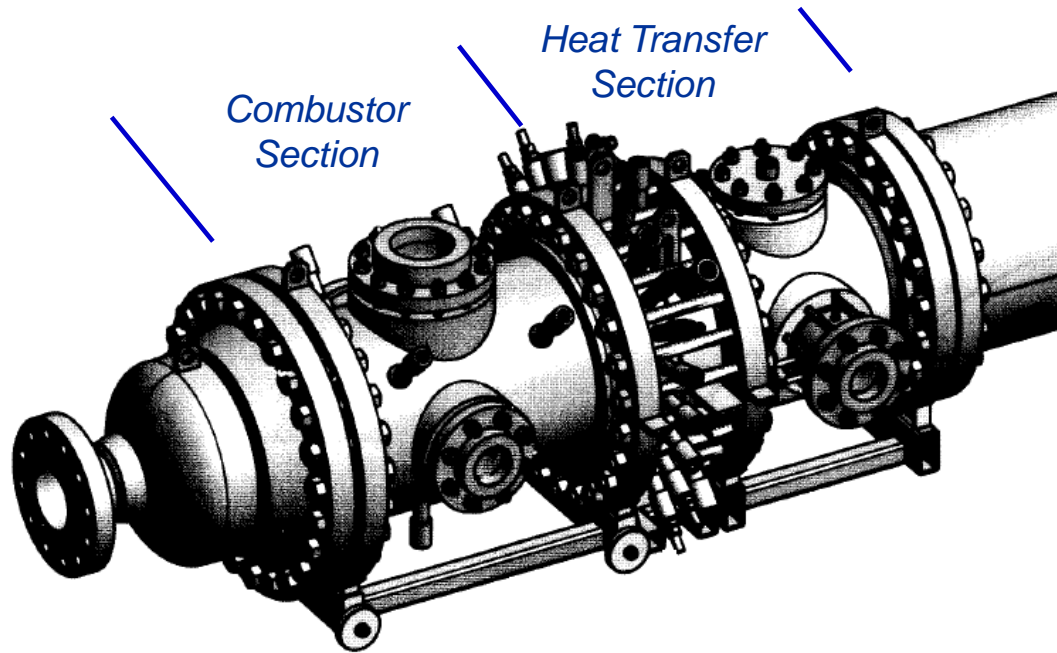


Optical viewports



Project Background

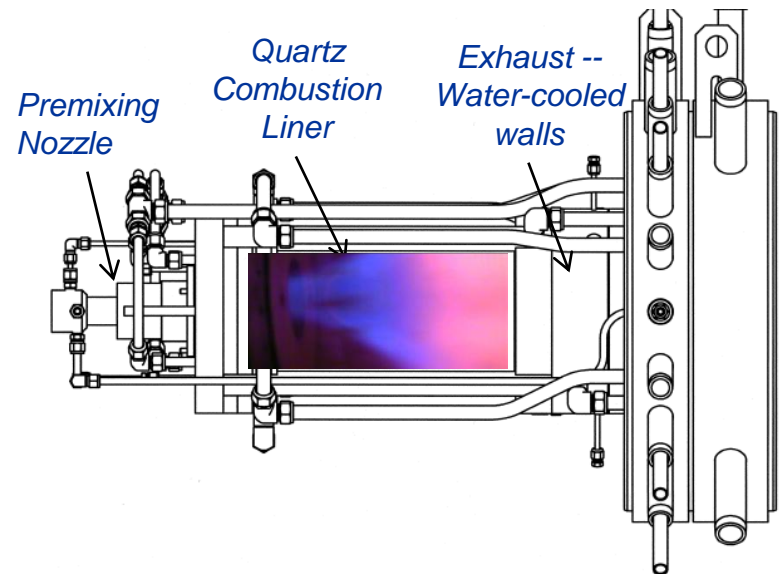
- **Modify an existing high pressure combustion rig**



- **Provide “realistic” hot gas path conditions for collaborative efforts**
- **“Proof-of-concept” testing for cooling and sensors**
 - Intermediate scale facility

Facility and Rig Capabilities

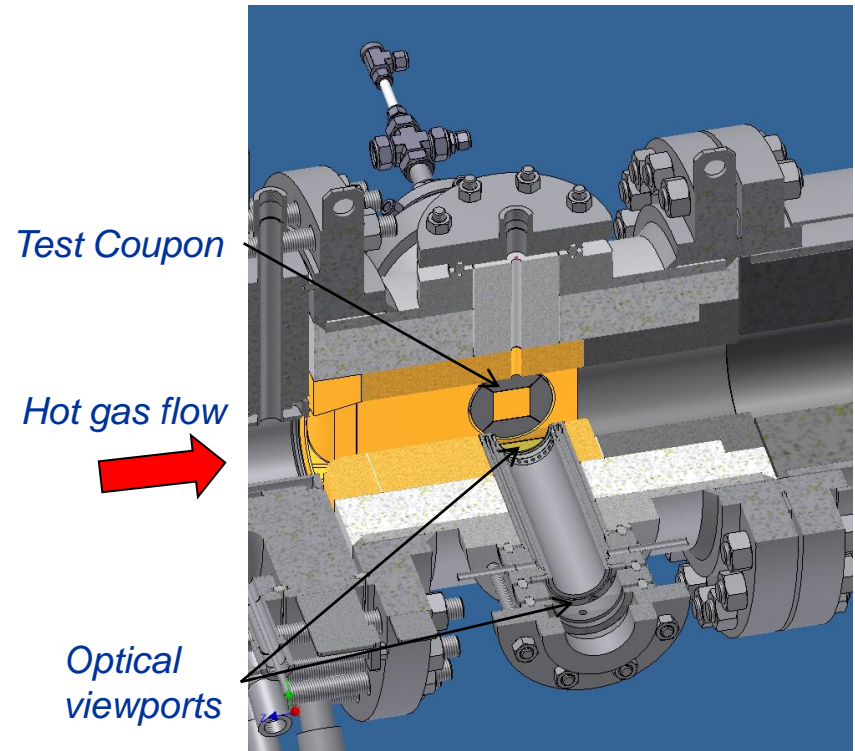
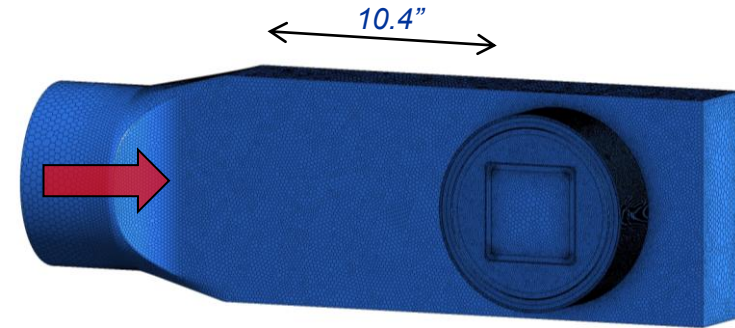
- **Facility capabilities**
 - 2 lb/s air flow @ 700 psi
 - 800-900 F air preheat (independent control)
- **Rig capabilities**
 - 2 lb/s air flow @ 10 atm
 - Max inlet air temperature (800F)
 - Natural gas or hydrogen fuels
- **Combustor design**
 - Swirl-stabilized
 - Lean premixed gaseous fuel
 - Diffusion pilot (12 jets)
 - Quartz combustor liner
 - No dilution cooling jets
 - No upstream film cooling



Experimental Setup

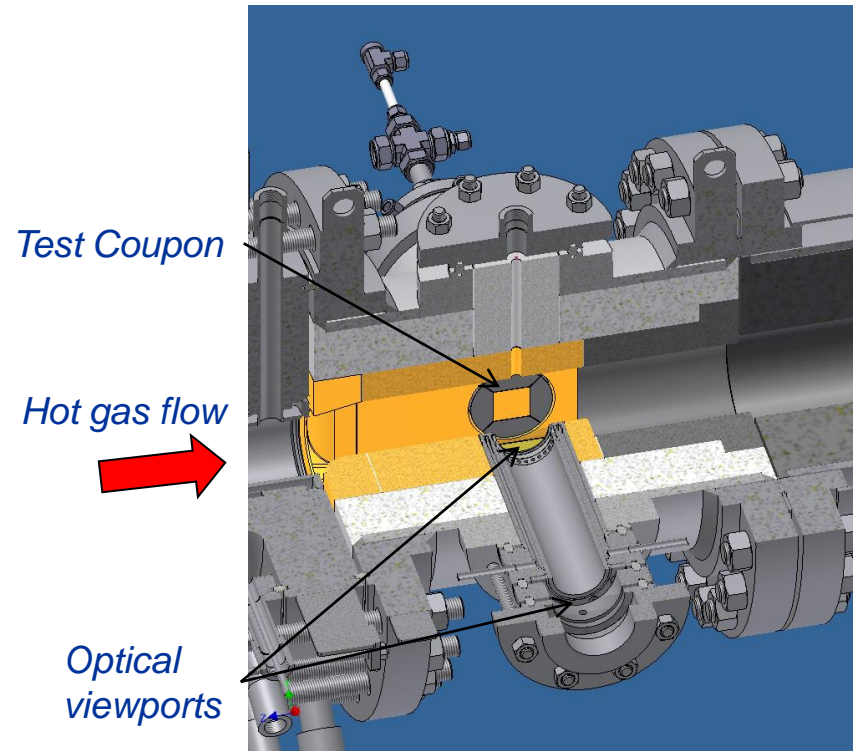
- **Heat transfer section**

- Refractory lined walls
- Transition to rectangular cross-section
 - Nominal 4" ID to 5"x 2" flow channel
- Test samples
 - Haynes 230 coupons
 - 2" x 2" x 0.25" thick
 - Flush with interior walls
 - With and without TBC's
- External viewport
 - Commercial quartz flange
- Internal viewport
 - 3" OD x 1/2" thick quartz
 - Flush with inner wall

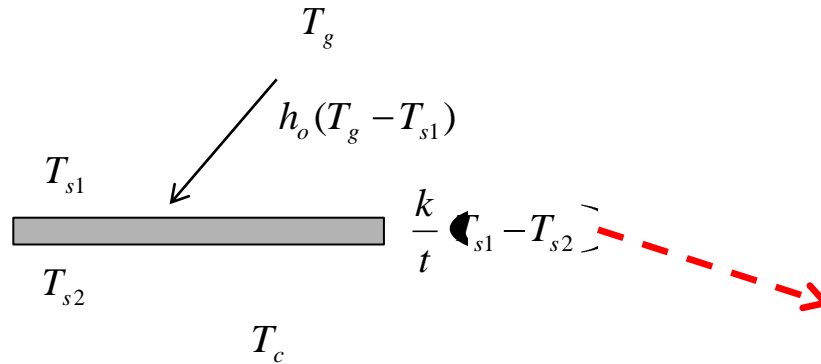


Independent Parameters

- **Operating pressure (1-10 atm)**
- **Hot gas temperature (1000C → ?)**
- **Hot gas path velocity**
 - Limited by flashback and blowoff in combustor premixer
- **Cooling air mass flowrate**
 - Blowing ratio
- **Test sample design**
 - Without TBC
 - With TBC
 - Film cooling design



Experimental Approach

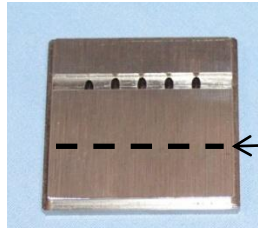


How are surface temperatures measured?

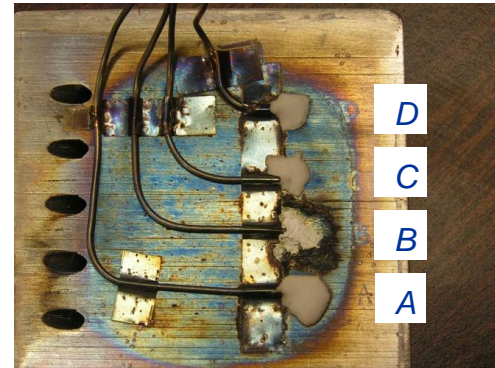
- **Measure q'' with and without coolant flows**

$$\Delta q_{red} = 1 - \frac{q''_{coolant}}{q''_{uncooled}}$$
- **Calculate heat flux**
 – $q'' = k \Delta T / \Delta t$
- **Gas temperature is measured using TC**
- **Calculate heat transfer coefficient, h_o**
 – $h_o = q'' / (T_g - T_{s1})$

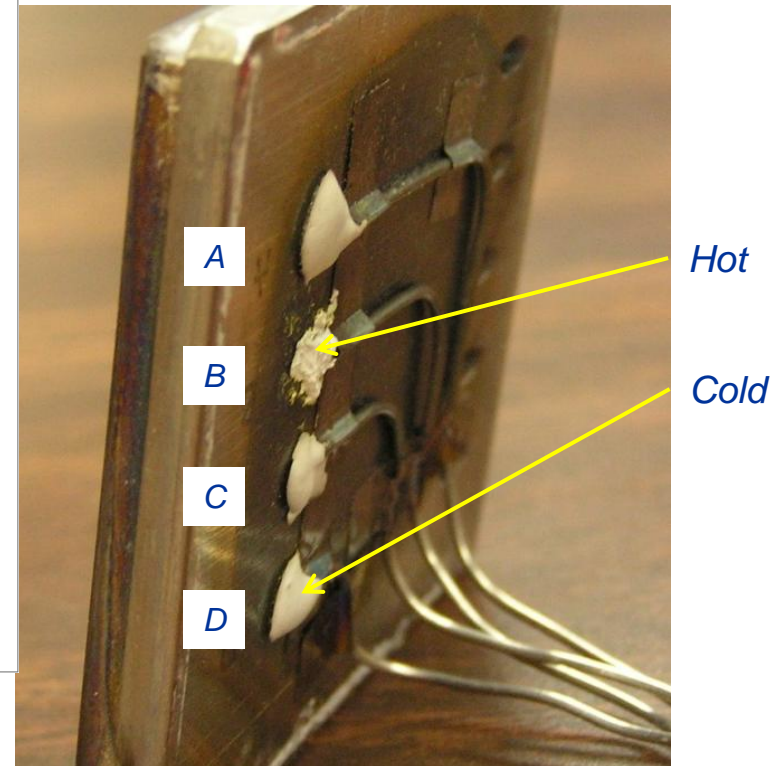
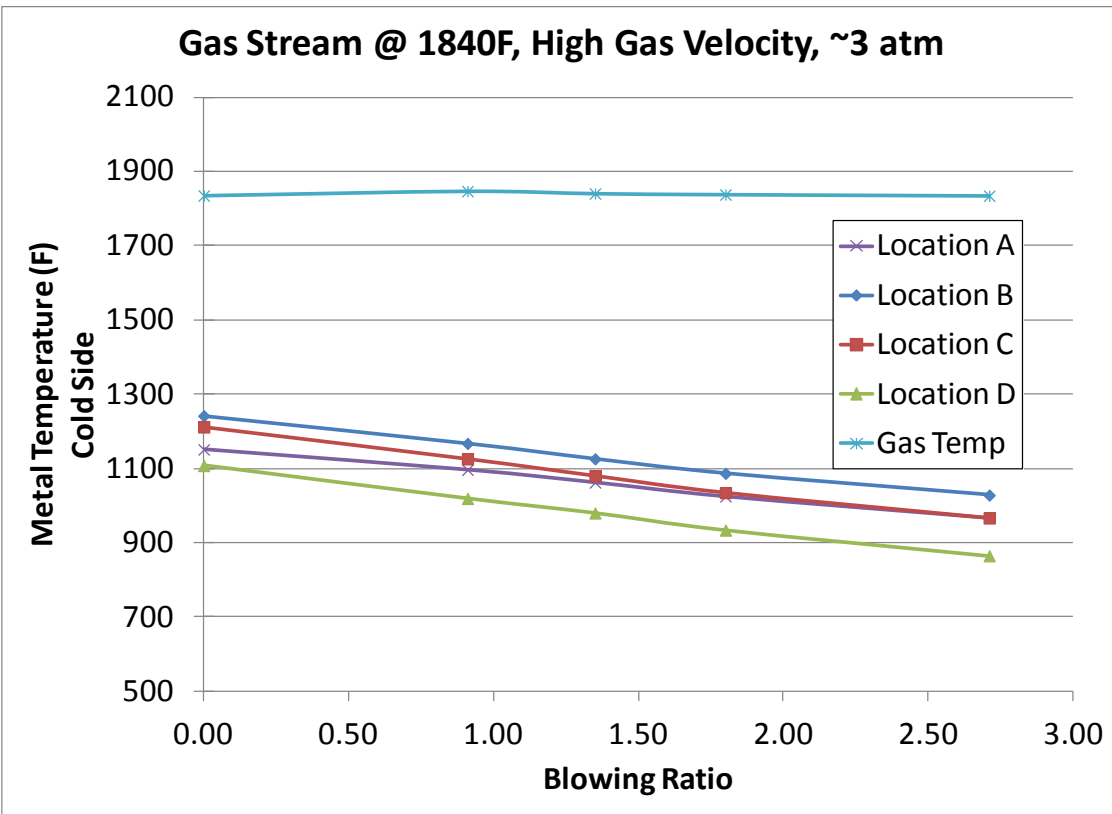
Cold Side Temperature Measurements



Approx. 2 hole diameters from trailing edge of hole

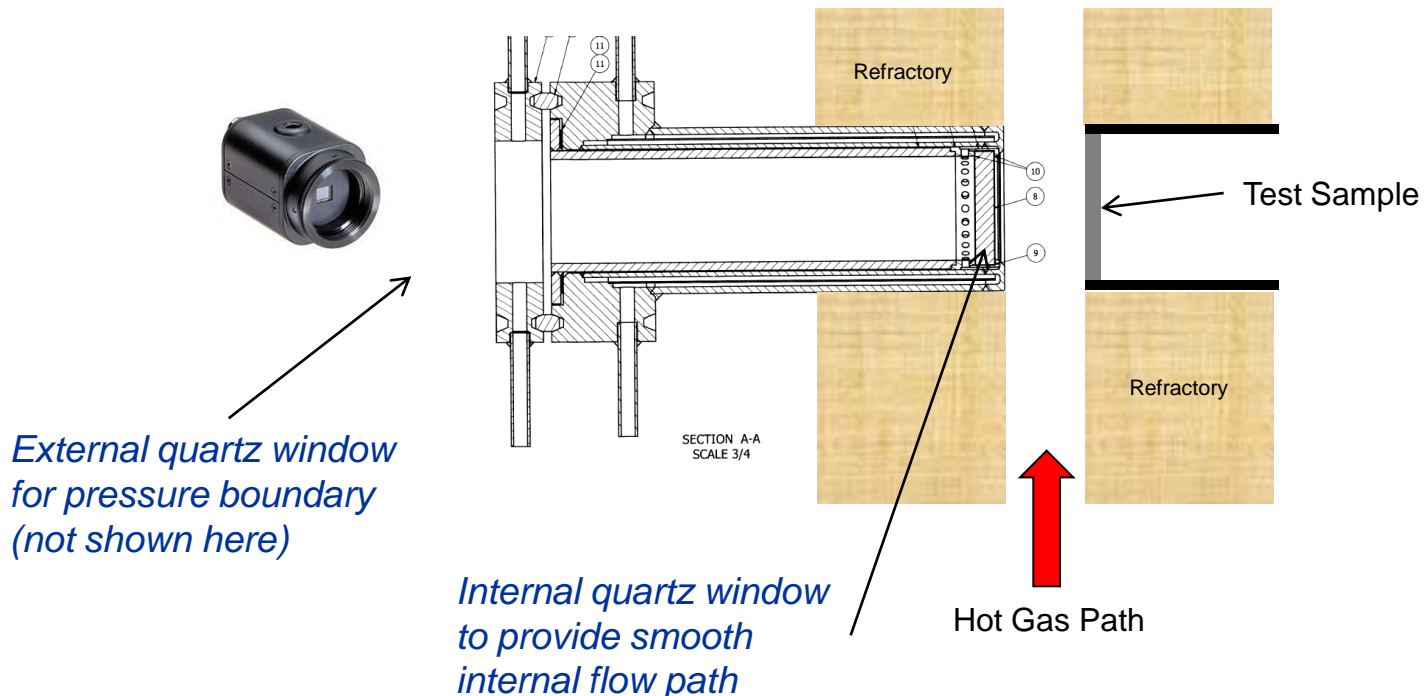


Gas Stream @ 1840F, High Gas Velocity, ~3 atm



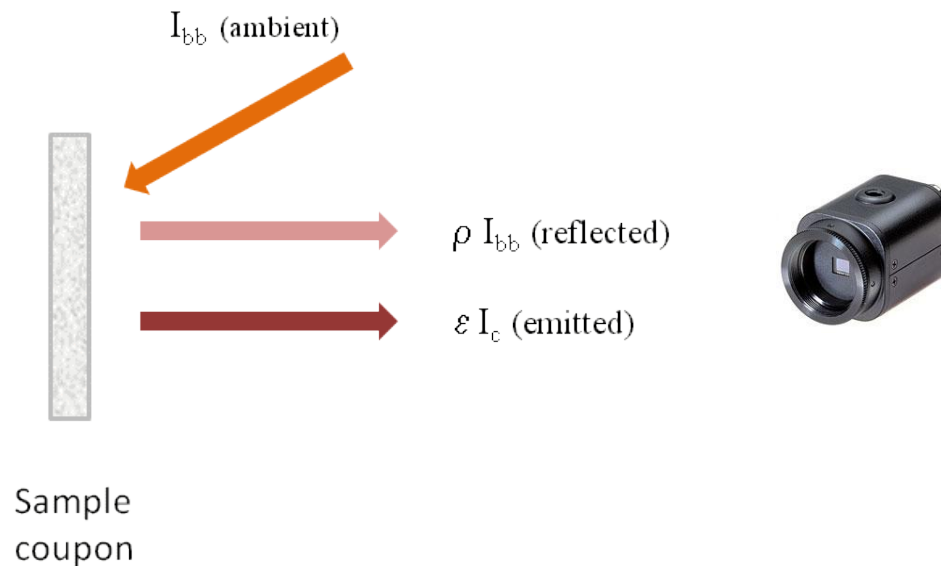
Surface Temperature Measurement Approach

- **Emissivity Corrected Pyrometer**
 - Measures surface emissivity @ 905nm
- **B&W CCD camera**
 - Bandpass (900nm) filter
 - Calibrated against BB source

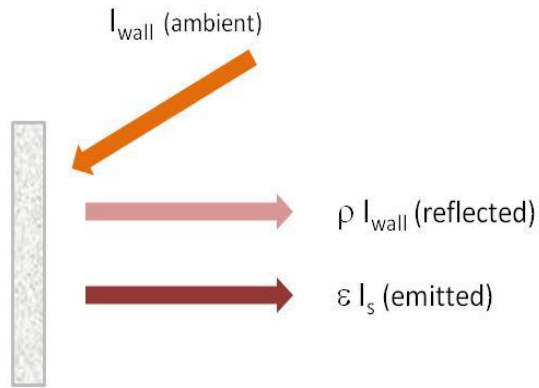


Hot Surface Temperature Measurement Issues

- **Optical approach cannot differentiate between reflected and emitted photons**
 - Uncertainty analysis to understand impact
 - Design of a multi-color probe in progress (Apogee Scientific)
 - Developing other options too!



Uncertainty Analysis



- Measured signal is sum of reflected and emitted radiation

$$I_{meas} = (1 - \varepsilon) \cdot I_{wall} + \varepsilon \cdot I_s$$

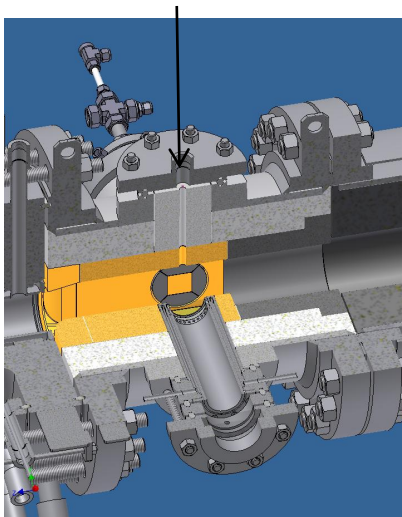
- In this simplified approach, assume aerothermal module interior (except test coupon) behaves as blackbody. With Wien's approx. to Planck's law -

$$I_i = \frac{2 \cdot h \cdot c^2}{\lambda^5} \cdot e^{\left(\frac{-C_2}{\lambda \cdot T_i} \right)}$$

- The sample-coupon temperature can then be written:

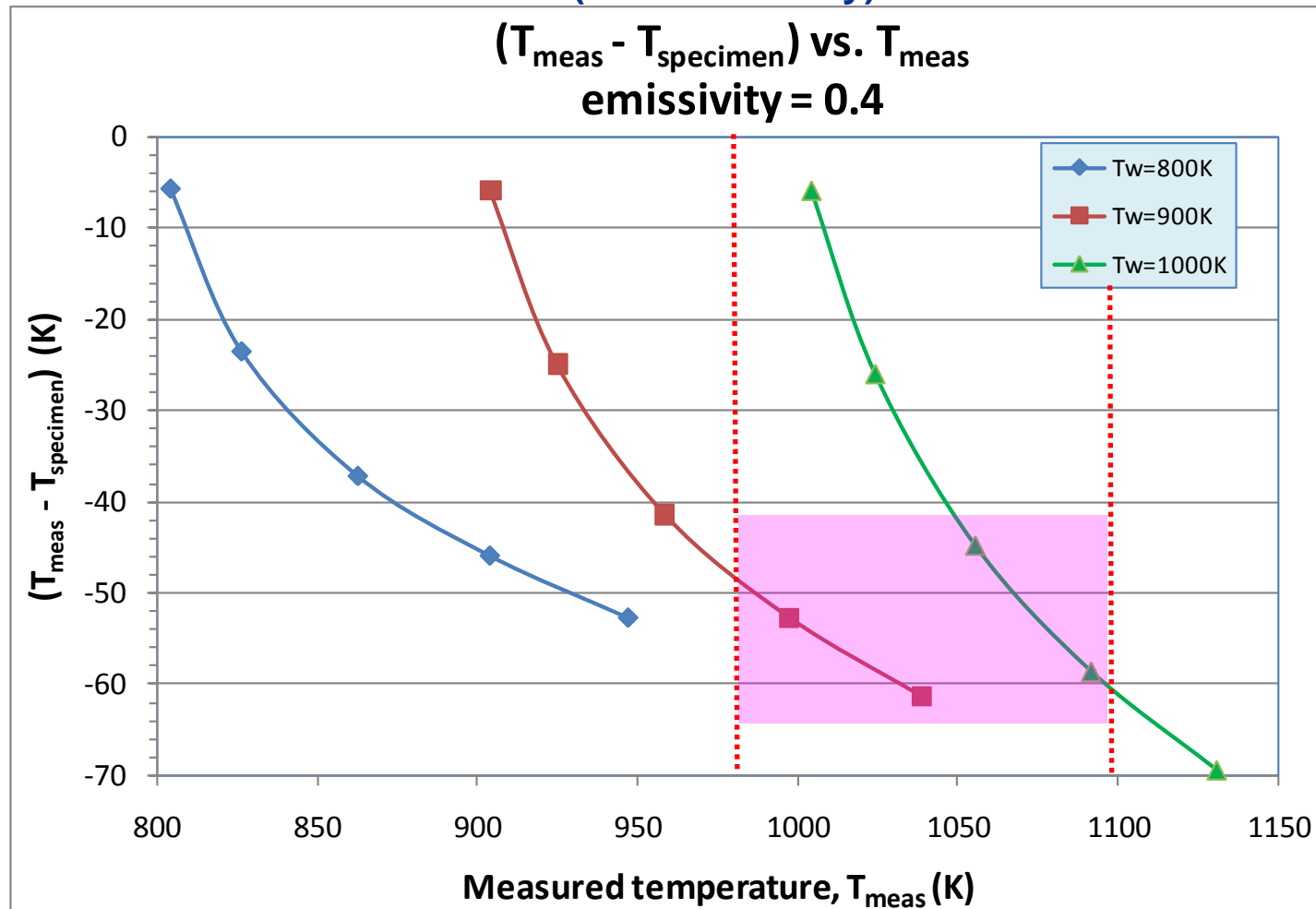
$$T_s = \frac{-C_2}{\lambda} \cdot \frac{1}{\ln \left[\frac{1}{\varepsilon} \cdot e^{\frac{-C_2}{\lambda} \cdot \frac{1}{T_{meas}}} - \left(\frac{1 - \varepsilon}{\varepsilon} \right) \cdot e^{\frac{-C_2}{\lambda} \cdot \frac{1}{T_{wall}}} \right]}$$

Multiple TC's



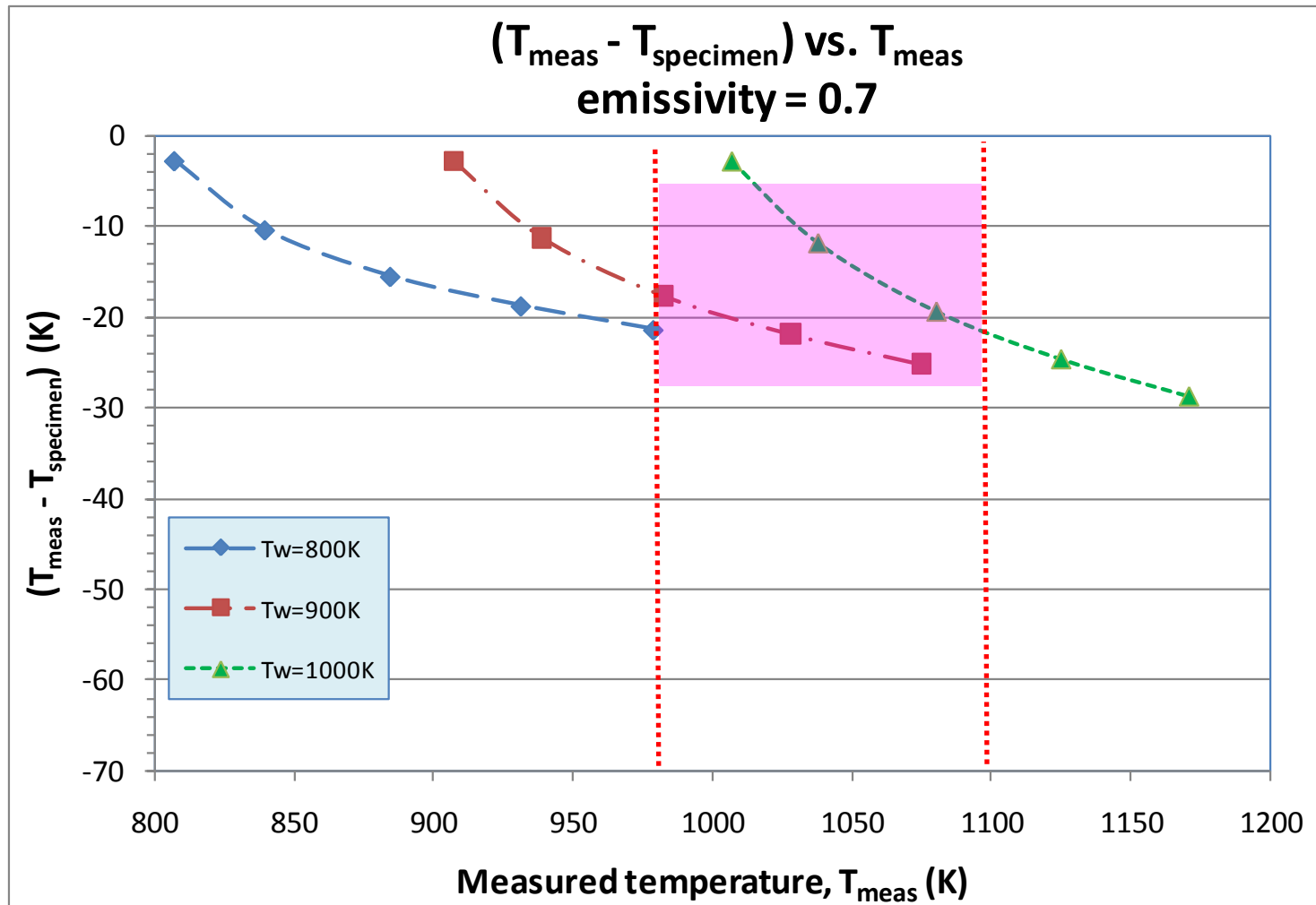
Estimated Bias In IR Mapping

(Low emissivity)



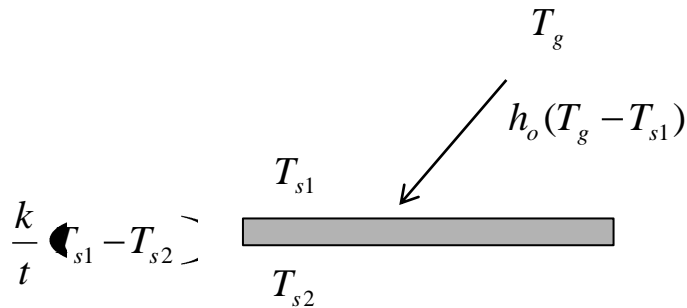
Estimated Bias In IR Mapping

(High emissivity)



Criteria for Uncertainty?

Variability of +/- 25% in the external convective heat transfer coefficient is acceptable



$$h_o(T_g - T_{s1}) = \frac{k}{t} (T_{s1} - T_{s2})$$

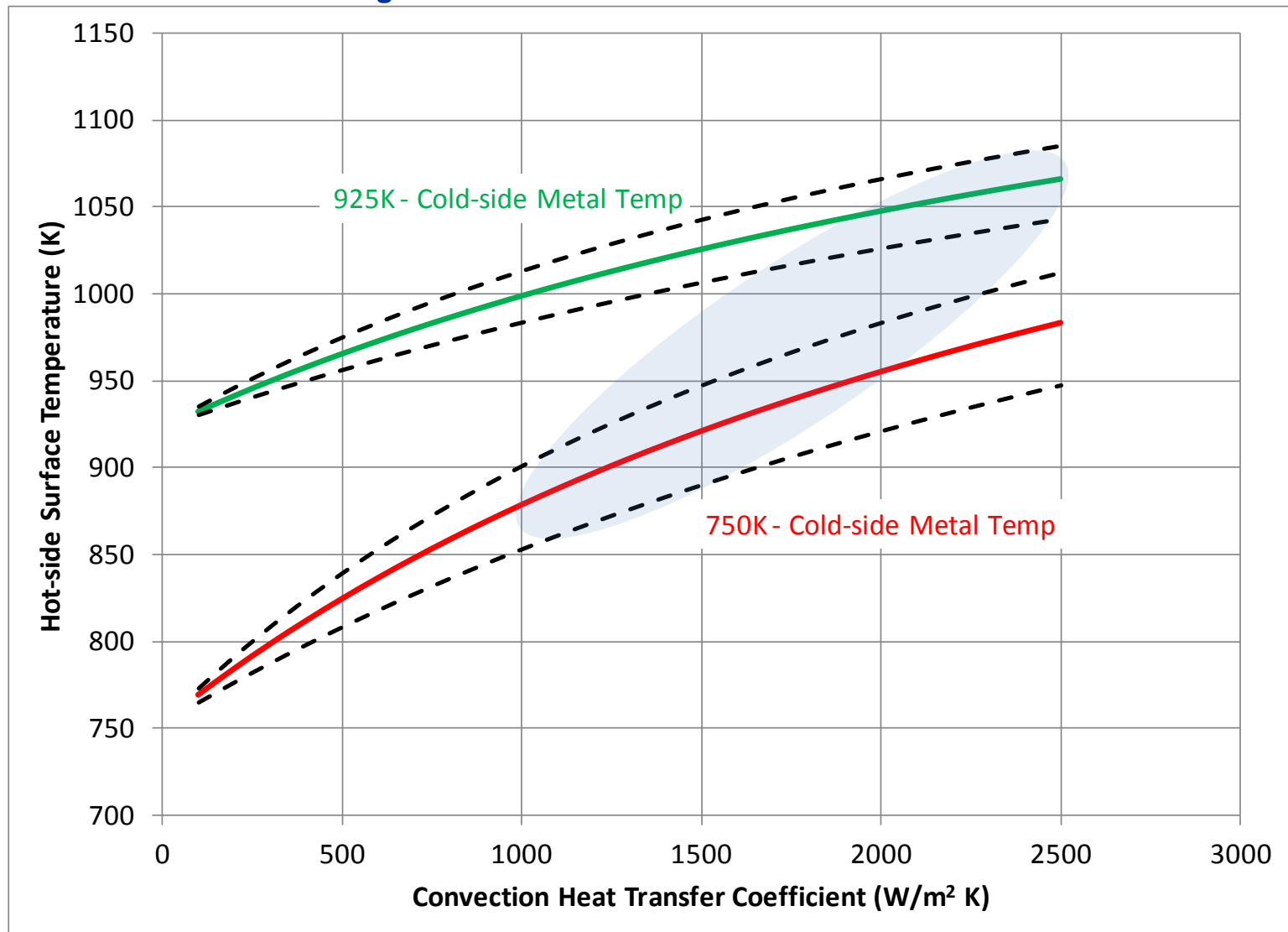
What is the acceptable level of variation in the hot-side surface temperature, T_{s1} ?

$$T_{s1} = \frac{\frac{t \cdot h_o}{k} T_g + T_{s2}}{1 + \frac{t \cdot h_o}{k}}$$

Fixed: $t \cdot h_o$ Fixed: T_g Bounded by experimental observations: T_{s2}
 $K = k(T)$

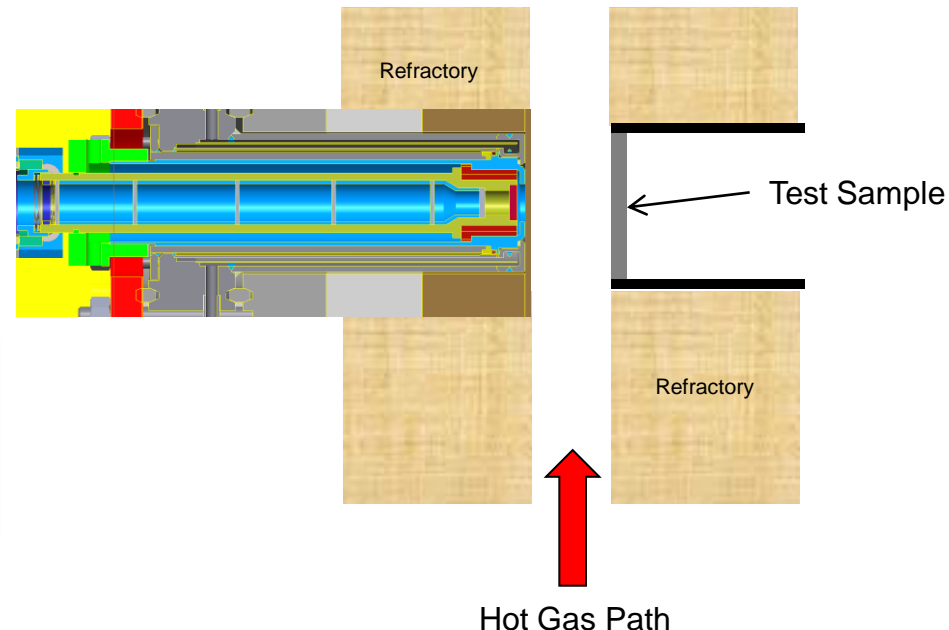
Criteria for Uncertainty?

($T_{\text{gas}}=1000\text{C}$, thickness = 0.25")



TIS -- An Accurate High Speed Alternative

- **Apogee Scientific (small business – Englewood, CO)**
 - High speed thermal imaging system (TIS) applied to combustion processes in reciprocating engines (*Automotive Engineering International*, April 2010 and *Automotive Design*, September 2010)
- **NETL/Apogee subcontract to design high-temperature thermal imaging system for NETL's Aerothermal Test Rig**
 - Probe design complete



Seeking industrial partnership to continue probe fabrication and testing at simulated turbine conditions

Lessons Learned

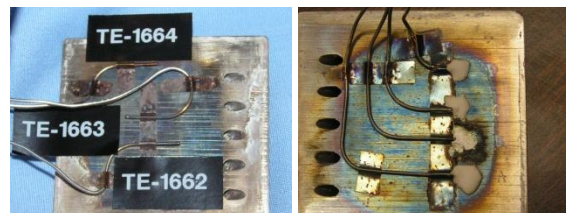
- **Quartz is a robust material, but . . .**
 - Contact stresses can ruin your day
 - Devitrification can ruin your measurement
- **Haynes 230 is a good material, but . . .**
 - Emissivity varies with oxidation
 - Thermal aging effects
 - Measured heat flux varies with exposure time
- **TBCs prevent oxidation but lower emissivity affects optical surface temp. measurements**
- **Attempts to maintain smooth interior walls were not successful**
- **Good surface temperature measurement is an art**



Before Test



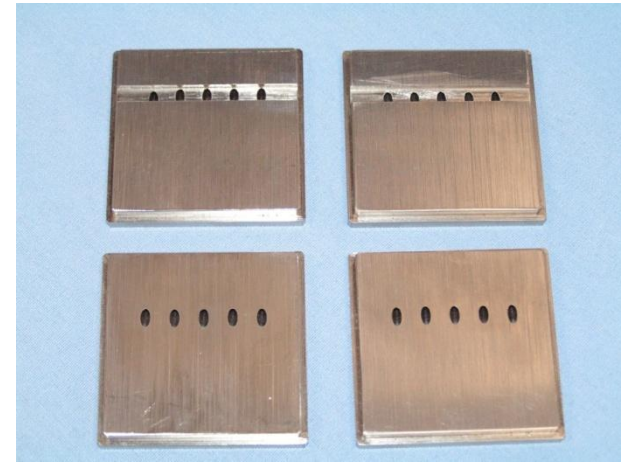
After Test



NETL/UNIVERSITY OF PITTSBURGH COLLABORATION ON FILM COOLING SAMPLES

NETL/Pitt Test Plan

- **Fuel composition (natural gas)**
- **Gas temperature**
 - Nominal 1000C (1835F)
 - Minimize devitrification of quartz window
- **Pressure**
 - 1 - 3 atm
- **Mass flowrates in hot gas path**
 - $15,000 < Re_L < 60,000$
- **Coolant (air) flowrate**
 - Blowing ratio (0-3)



No	Coupon Design	With TBC	Without TBC
1	Flat Plate		√
2	Cylindrical Hole	√	
3			√
4	Cylindrical Hole in Trench	√	
5			√
6	Shaped-Laidback Fan Hole		√

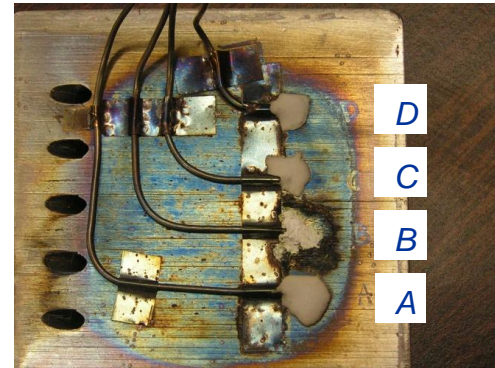
NETL/Pitt Collaboration Status Summary

- **Five film cooled coupons tested**
 - 30 degree cylindrical hole design
 - Haynes 230
 - Haynes 230/MCrAlY/8-YSZ TBC
 - Trench with 30 degree holes
 - Haynes 230
 - Haynes 230/MCrAlY/8-YSZ TBC
 - Shaped hole @ 30 degrees
 - Haynes 230

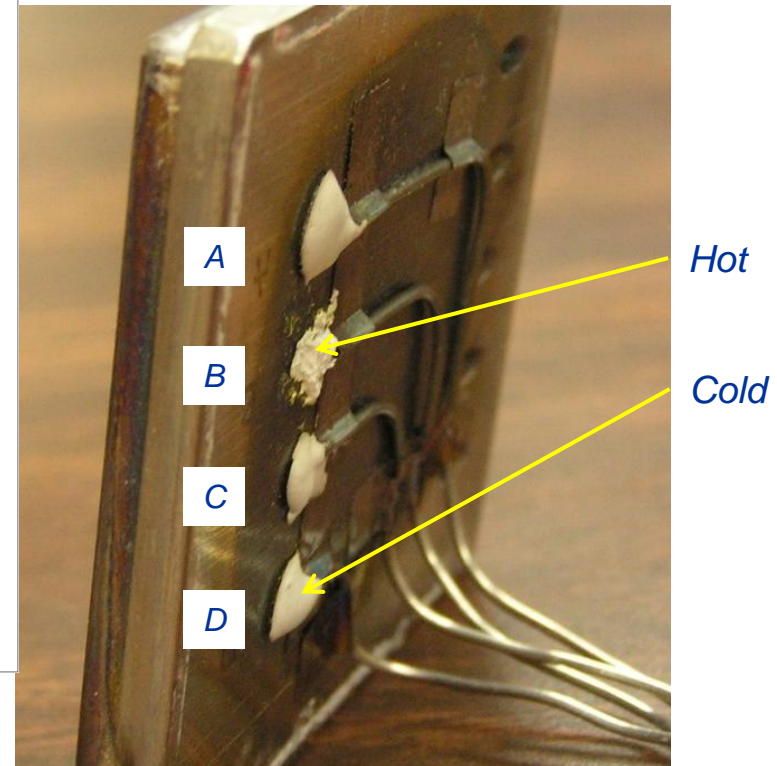
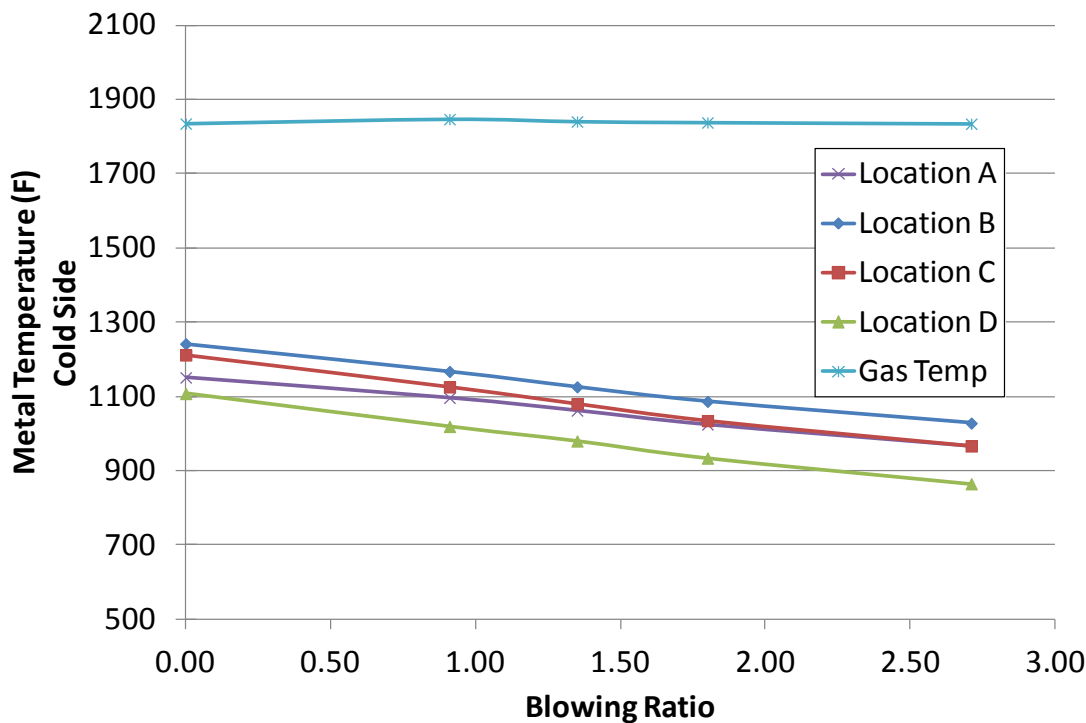


Trench Coupon – Backside TC

Approx. 2 hole diameters from trailing edge of hole



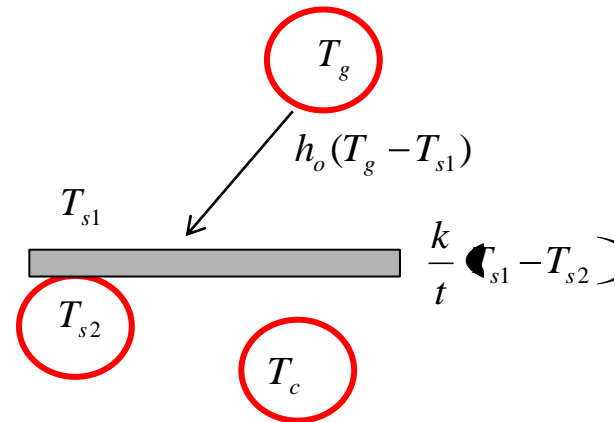
Gas Stream @ 1840F, High Gas Velocity, ~3 atm



Preliminary Test Results

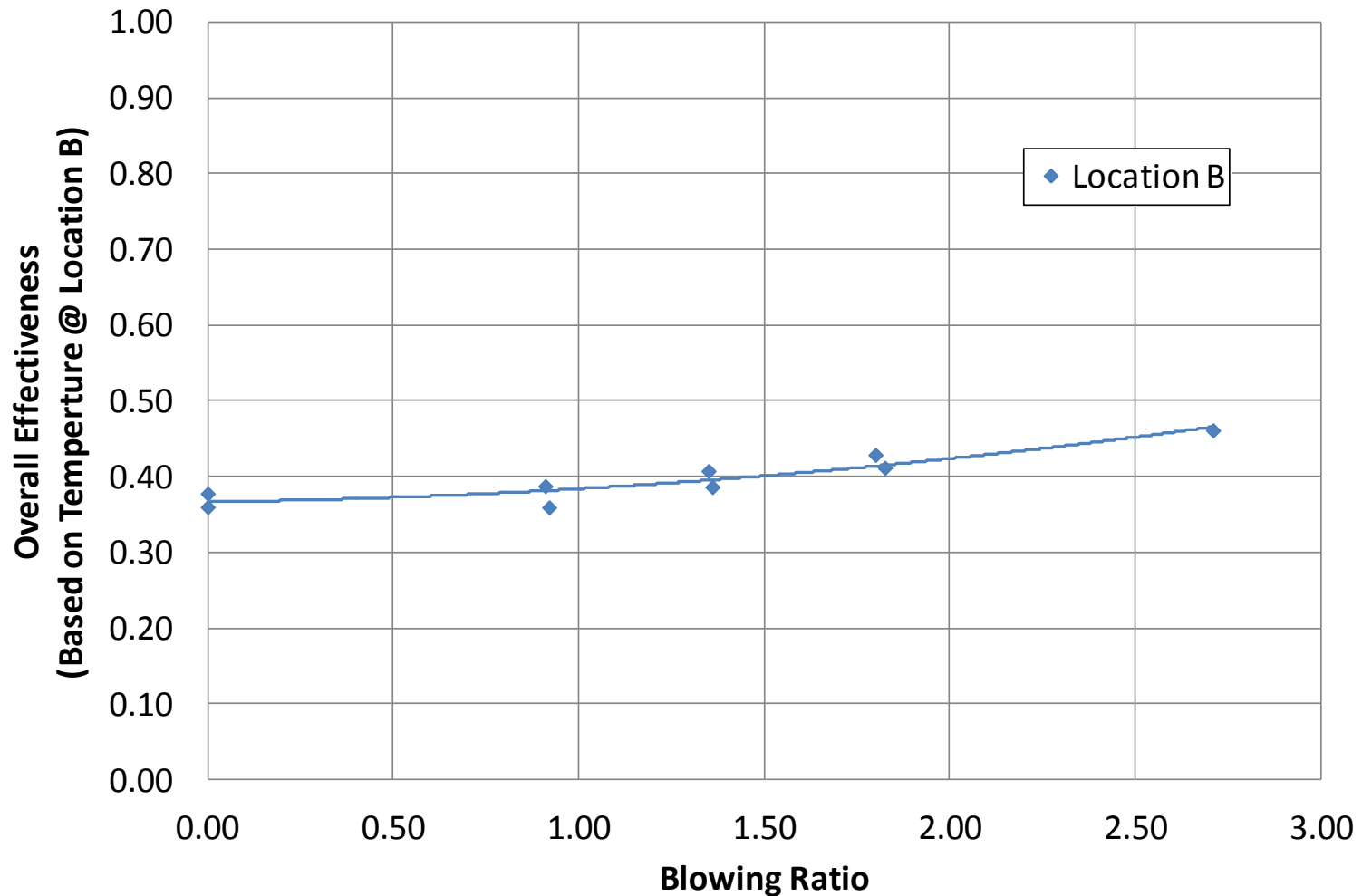
- Welded TC leads seem to provide highest temperature
- Performance indicator
 - Overall effectiveness

$$\phi = \frac{T_g - T_{s2}}{T_g - T_c}$$

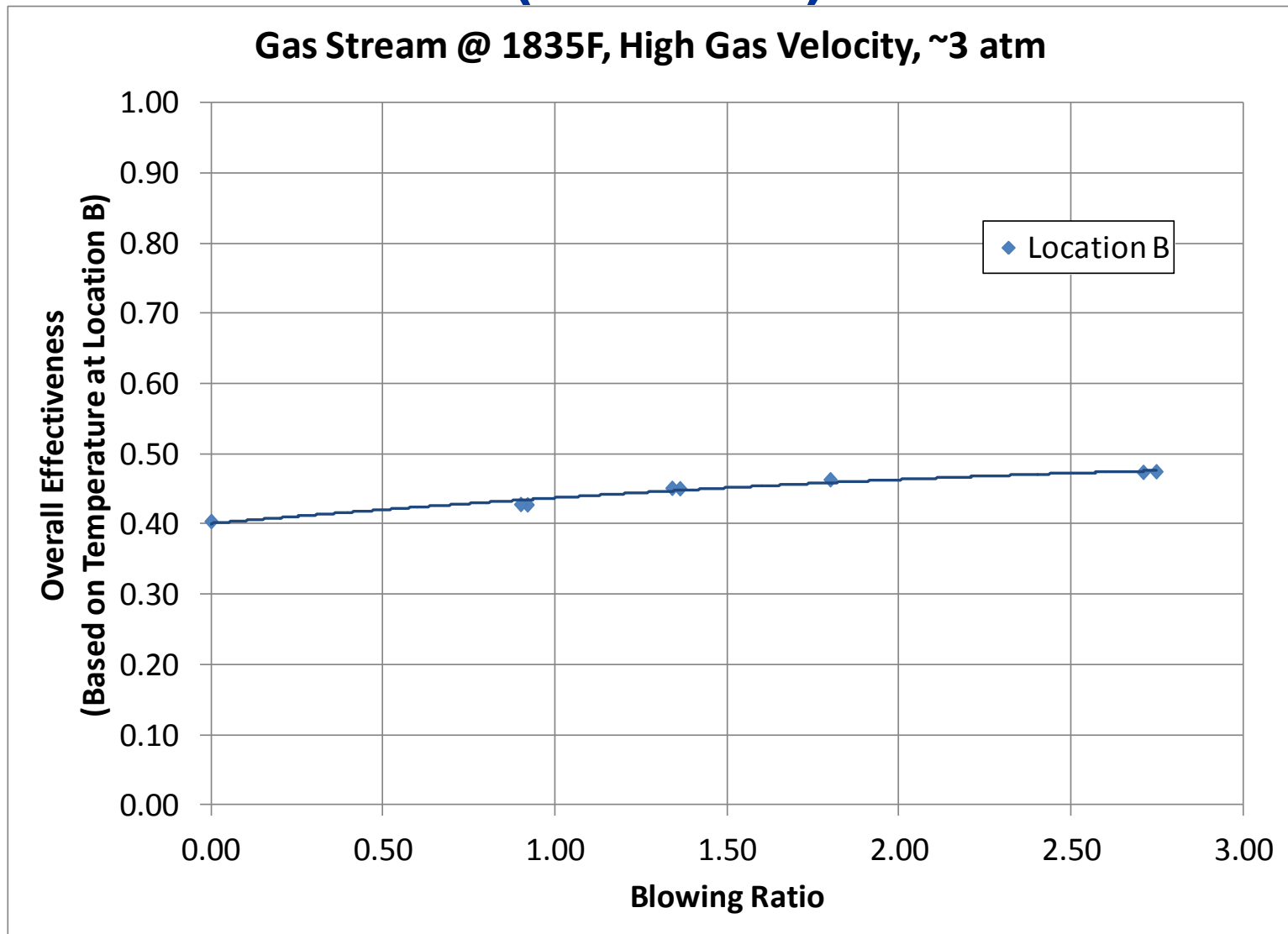


Trench Coupon – Repeatability (No TBC)

Gas Stream @ 1840F, High Gas Velocity, ~3 atm

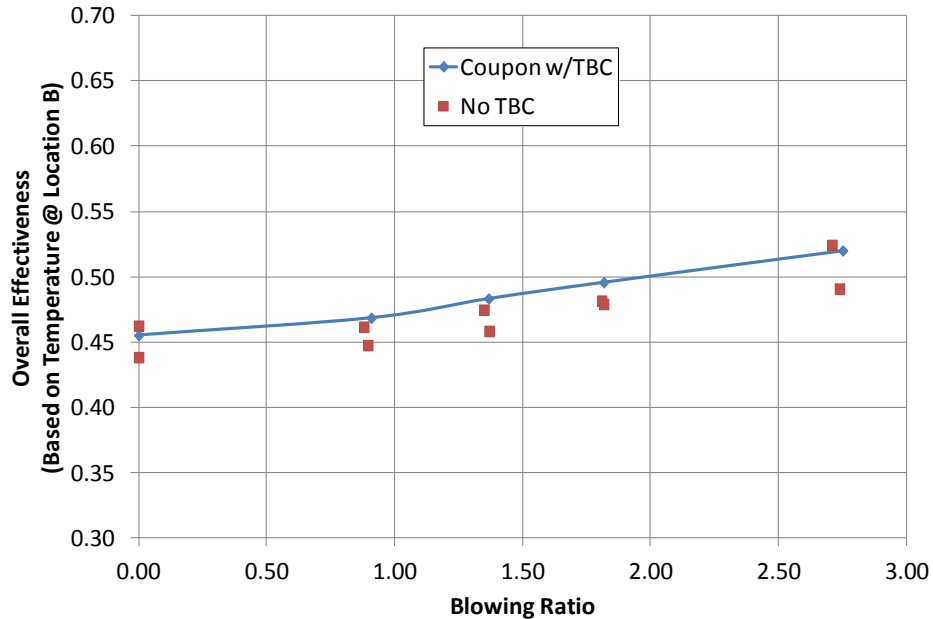


30 deg Cylindrical Hole Coupon – Repeatability (with TBC)

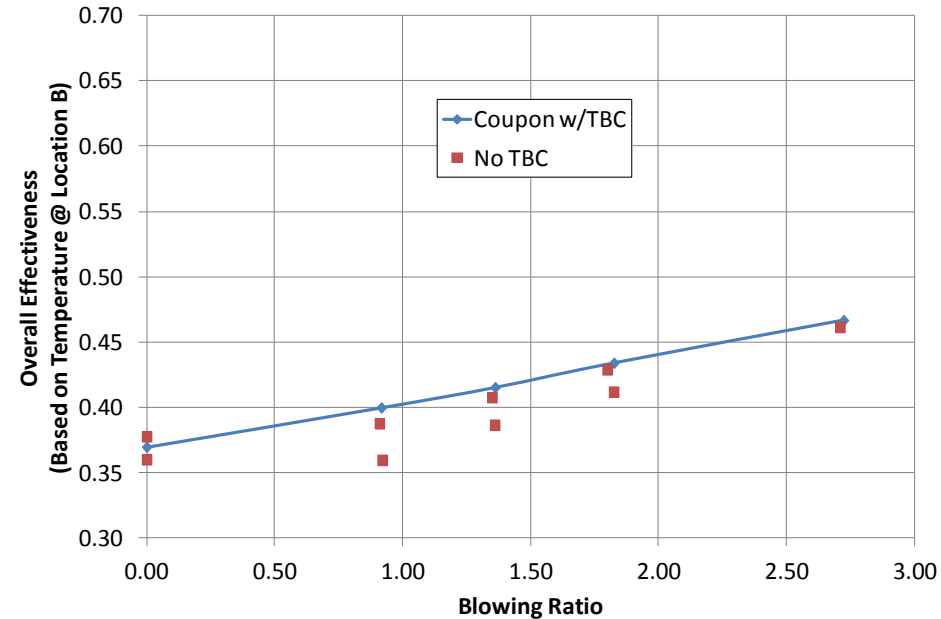


TBC Effects - Trench Coupon

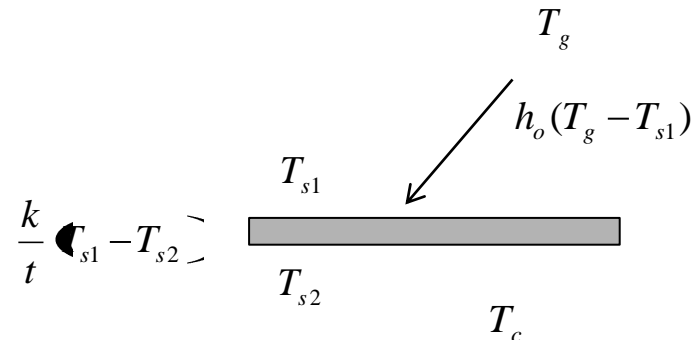
Gas Stream @ 1830F, High Gas Velocity, ~1 atm



Gas Stream @ 1830F, High Gas Velocity, 3 atm

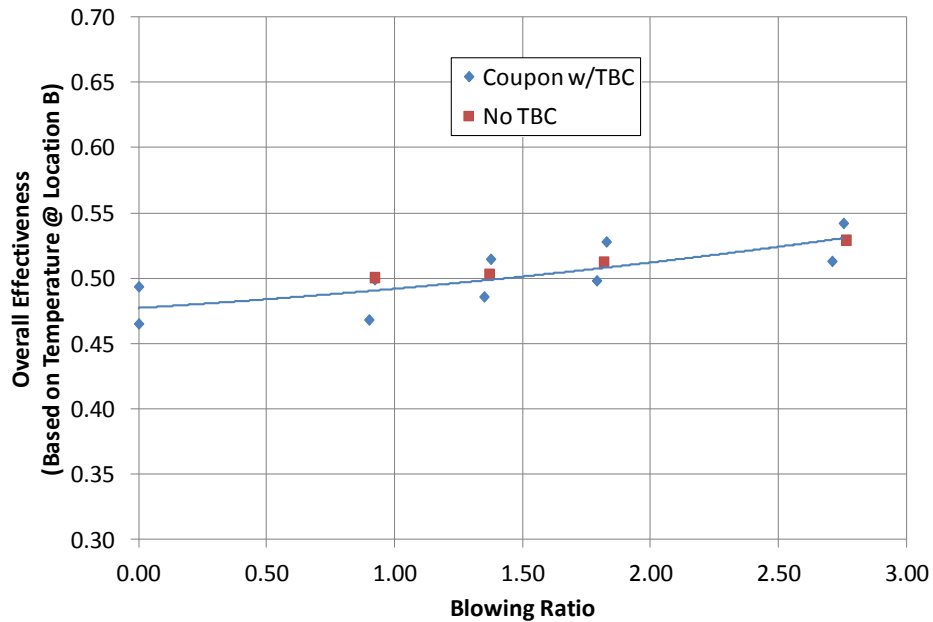


$$\phi = \frac{T_g - T_{s2}}{T_g - T_c}$$

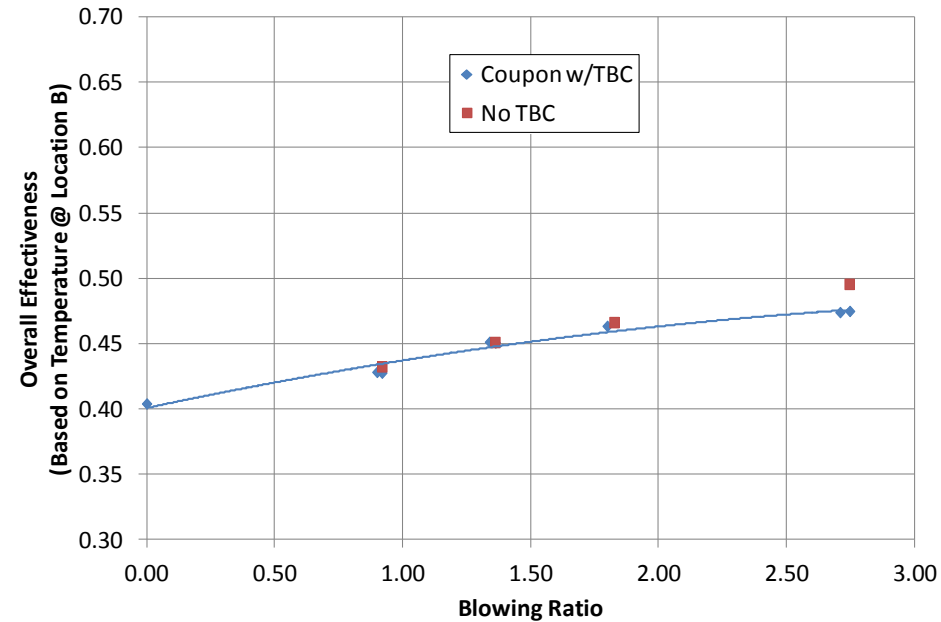


TBC Effects – Cylindrical Hole Coupon

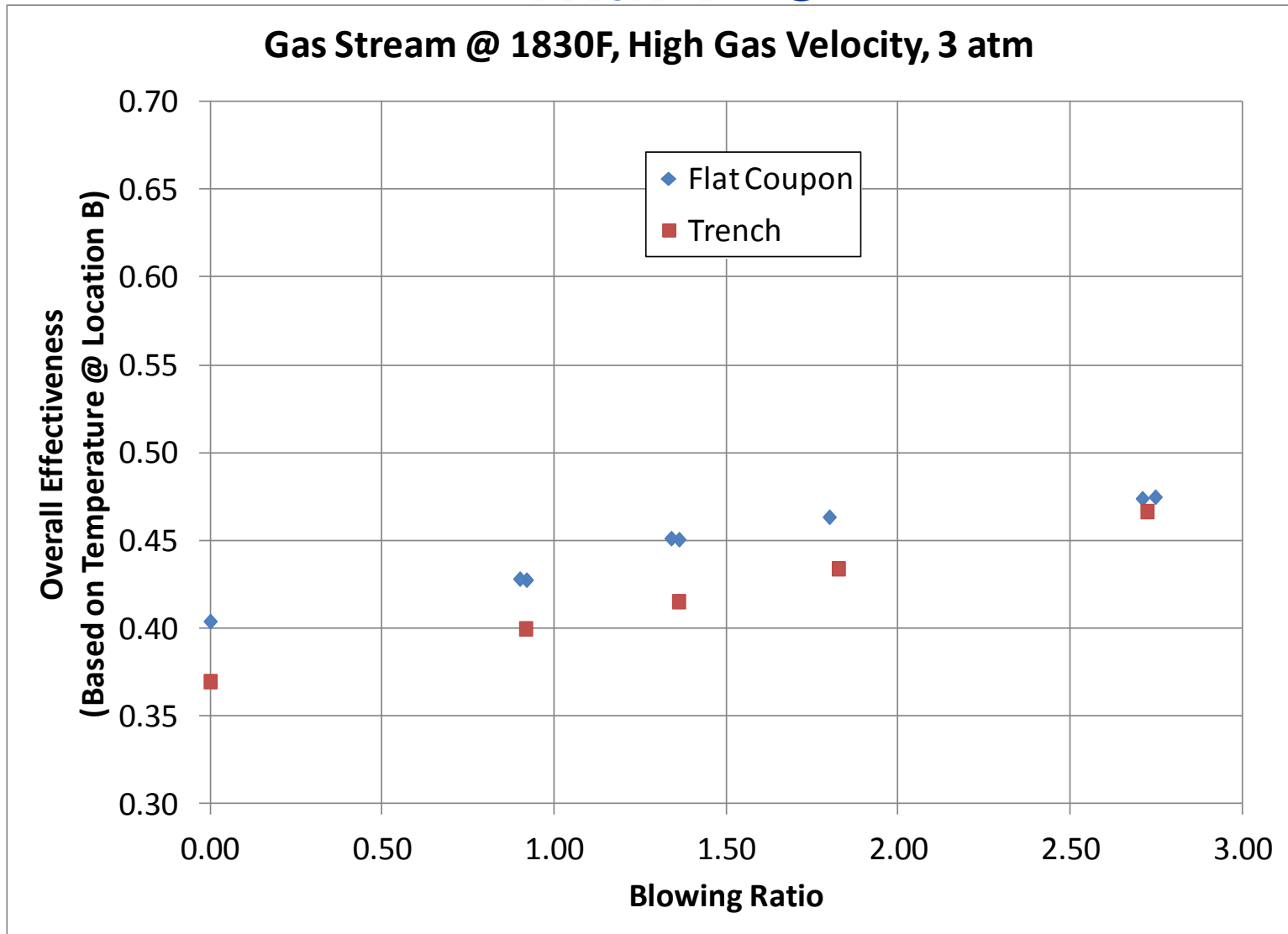
Gas Stream @ 1830F, High Gas Velocity, ~1 atm



Gas Stream @ 1830F, High Gas Velocity, 3 atm

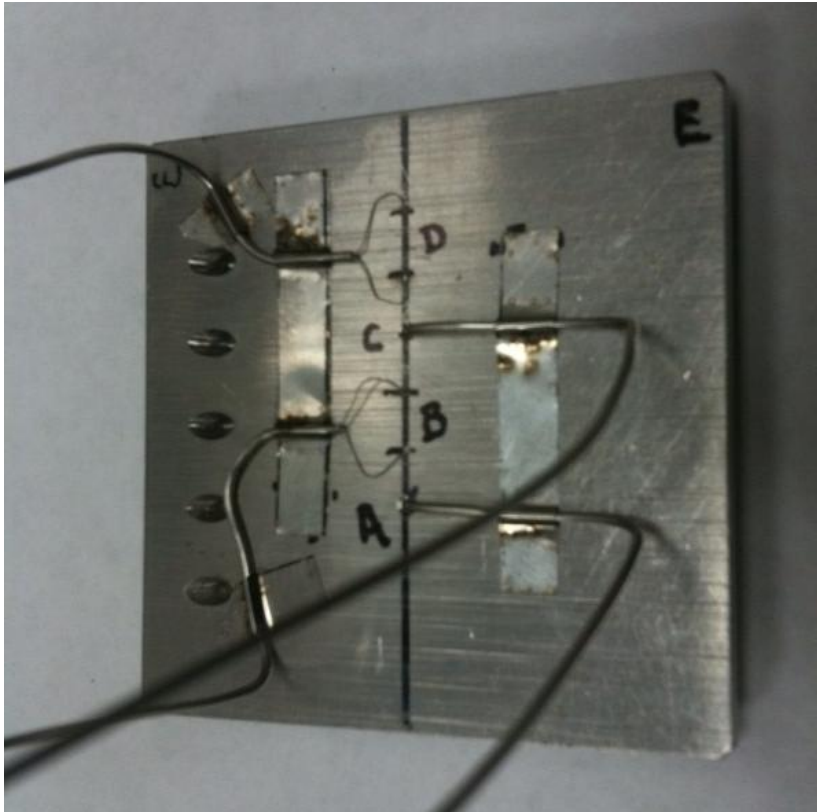


Trench Coupon vs Cylindrical Hole Coupon With TBC

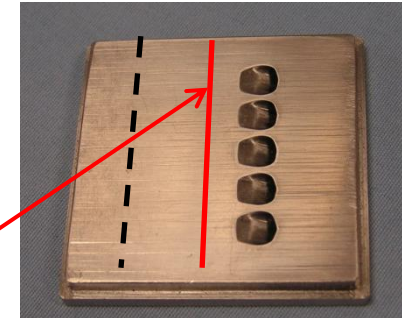


Fan-Shaped Hole Coupon

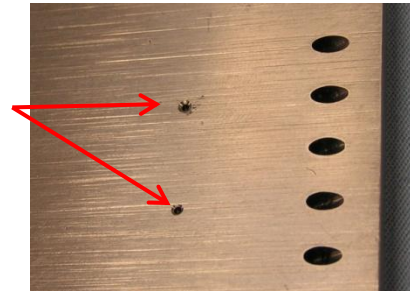
- TC attachment approach was changed
 - Two embedded TC's
 - Two TC's welded to surface



*New Line of
Backside TC
Locations*



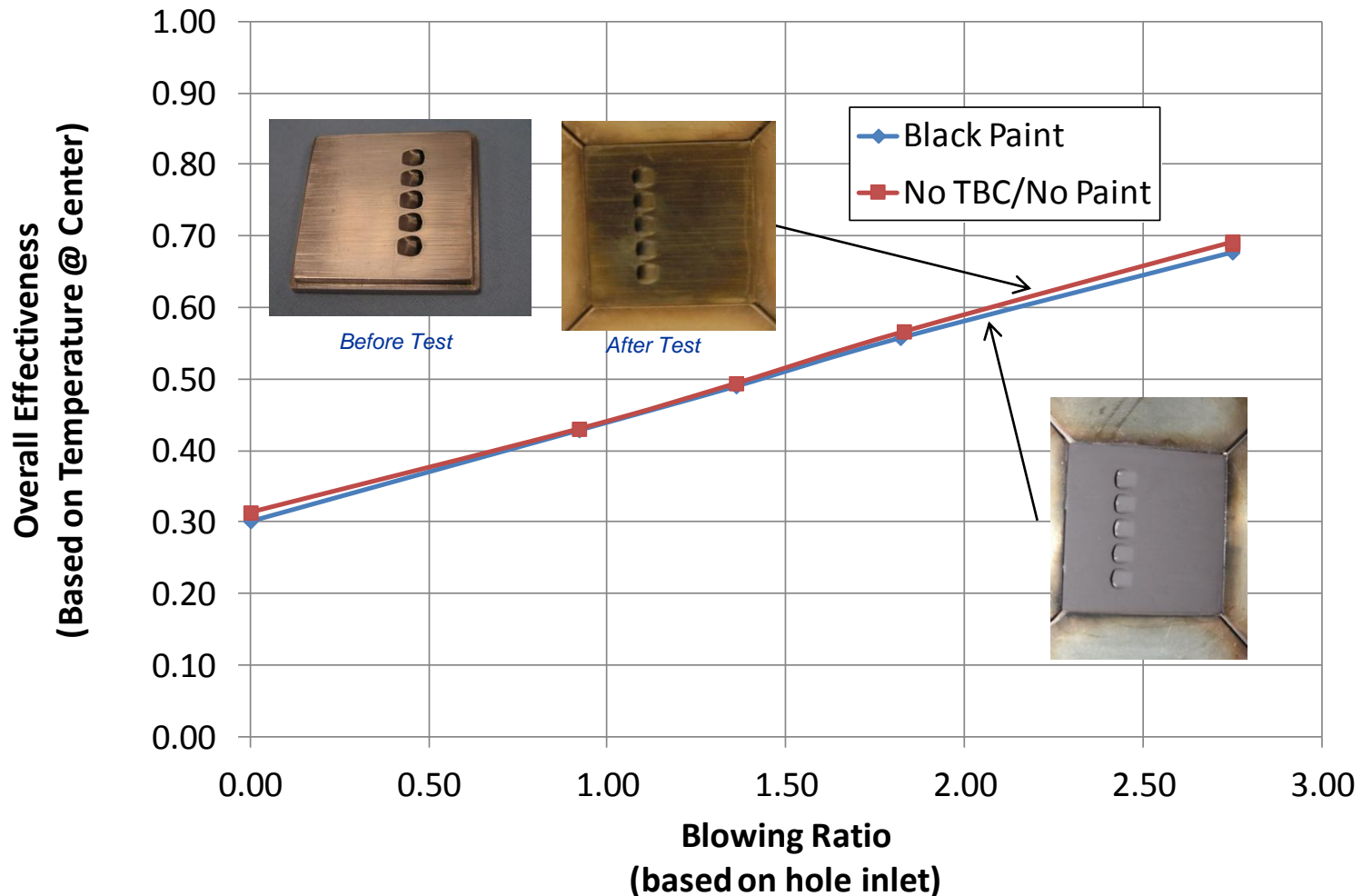
*TC's embedded
50% of thickness*



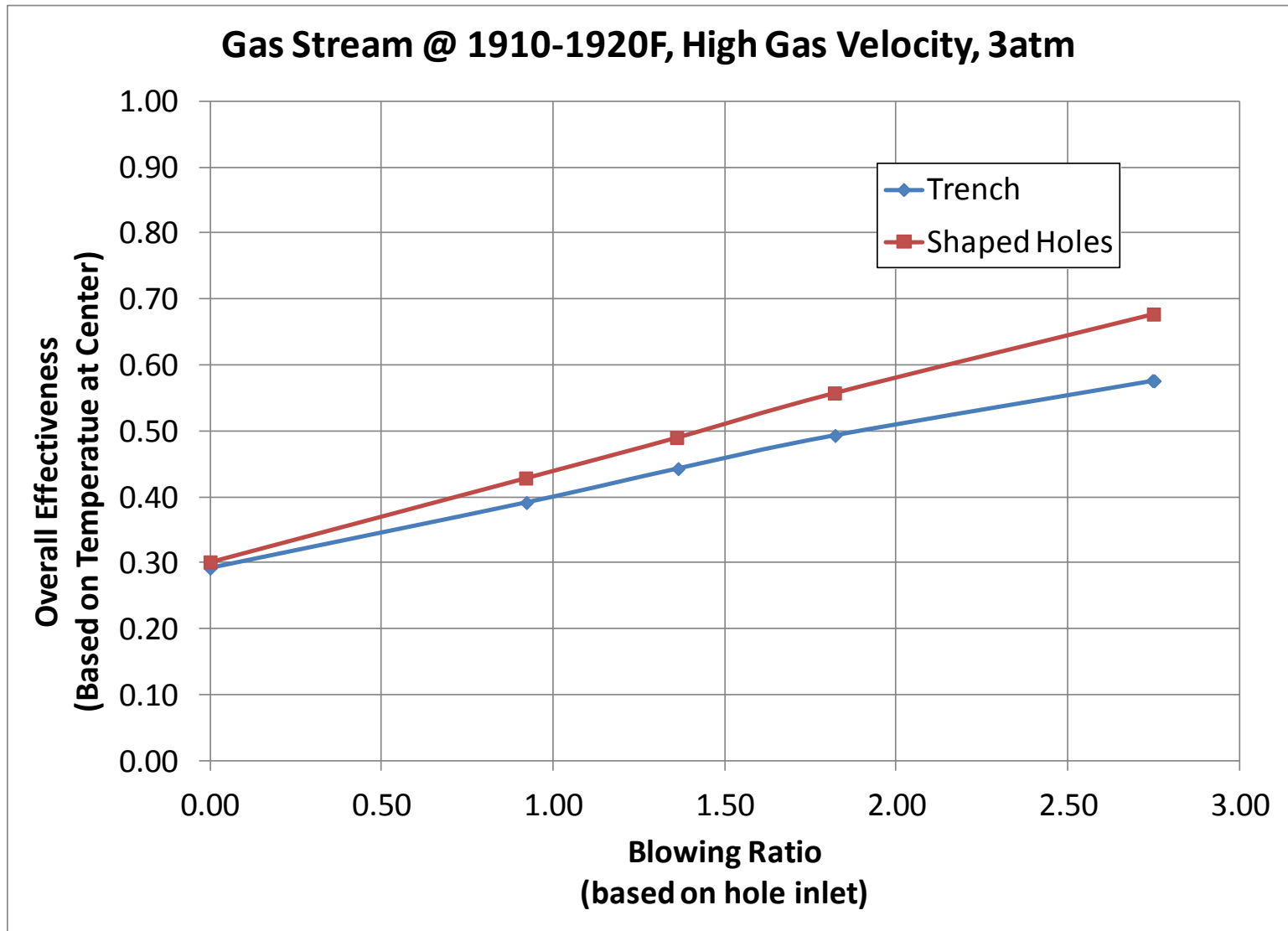
*TC's welded
equidistance
between embedded
TC's*

Fan-Shaped Hole Coupon (Painted Black vs As-Machined)

Gas Stream @ 1920-1940F, High Gas Velocity, 3atm



Trench Coupon vs Shaped Holes



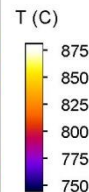
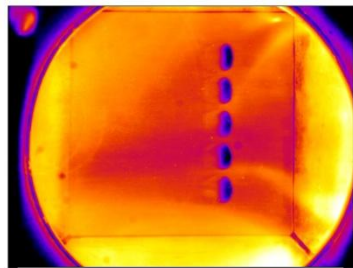
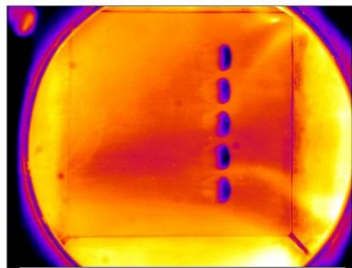
Thermal Images of Hot Side

(Coupons Painted Black; Function of Blowing Ratio, BR)

Shaped Hole Coupon

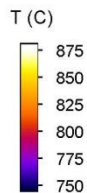
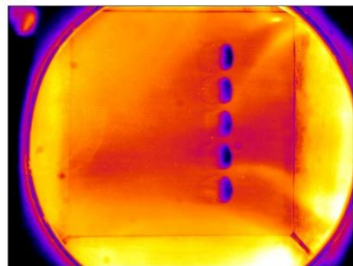
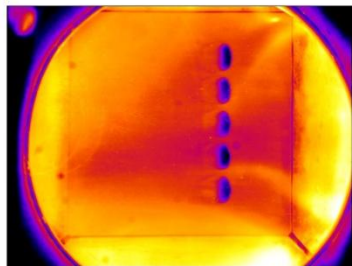
BR = 2.75

BR = 1.82

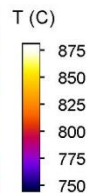
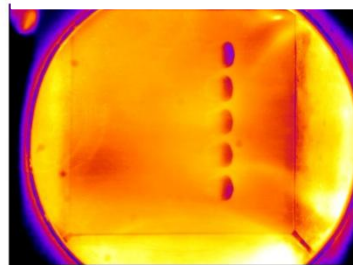


BR = 1.36

BR = 0.92



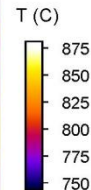
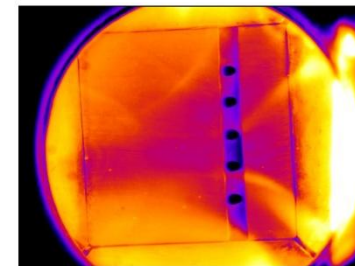
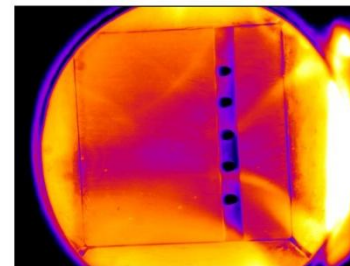
BR = 0



Trench Coupon

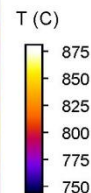
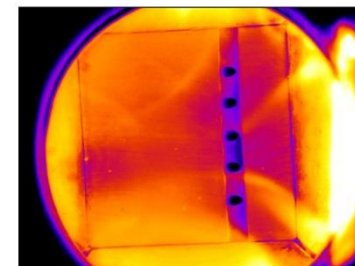
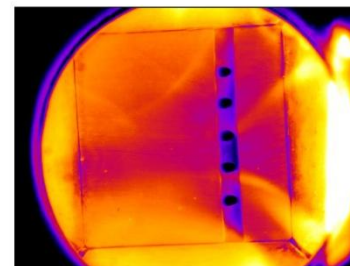
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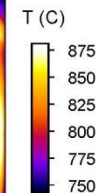
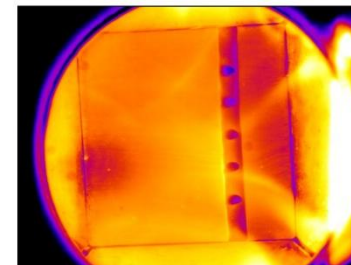


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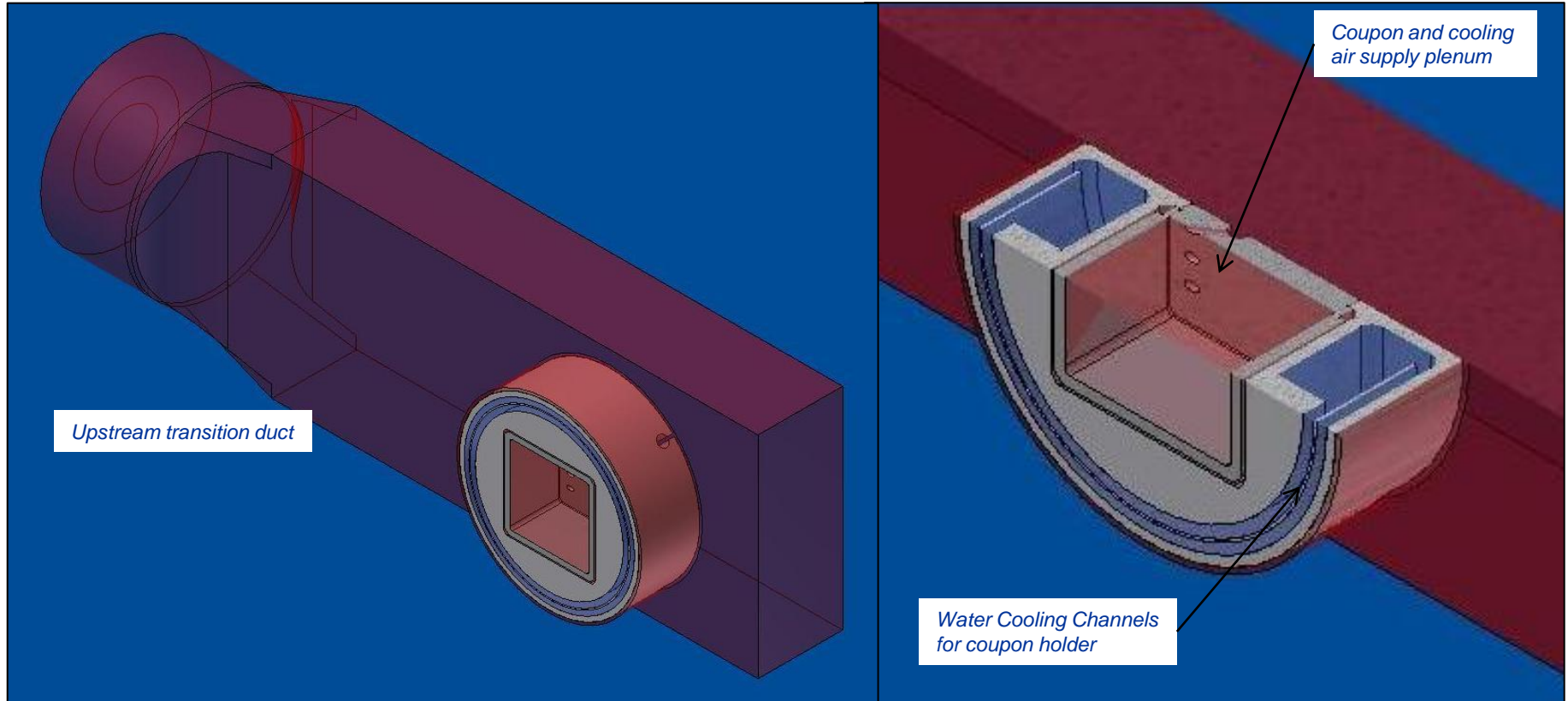
BR = 0



Current Status of Test Rig

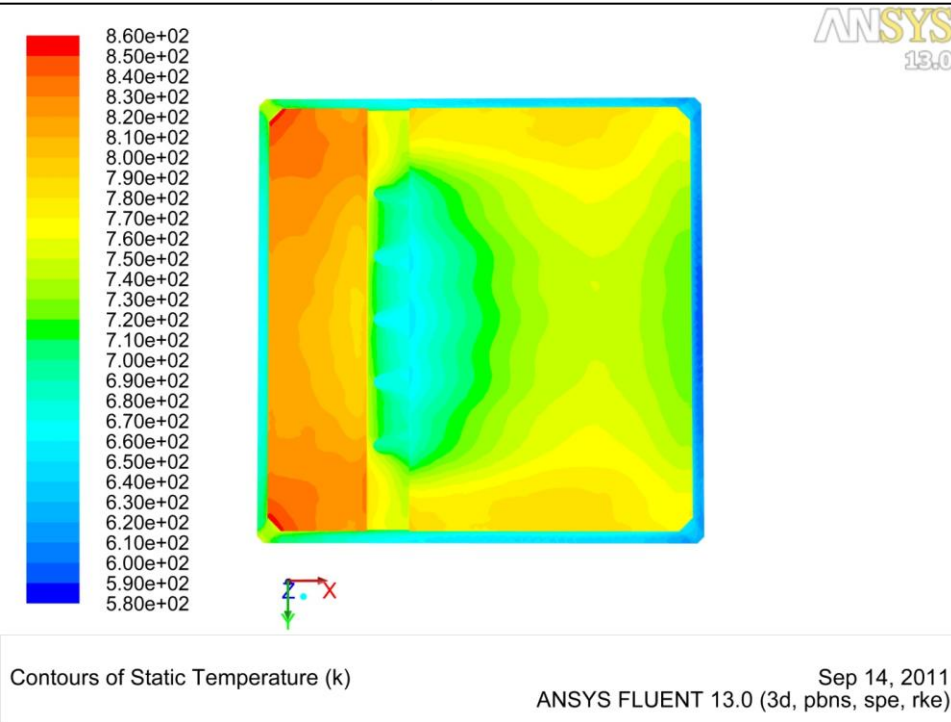
- **Preliminary film cooling coupon tests completed**
 - University of Pittsburgh
- **Develop In-situ LDV Capabilities**
 - All safety documentation has been addressed and permit to operate has been received
 - Laser safety enclosure designed, installed, and tested
 - Laser power supply failed during shakedown testing
- **Ash Deposition Studies (WVU)**
 - Particle seeder calibration complete
 - Hot shakedown testing started

Plans for FY12 to Include CFD Effort Using Commercial Software

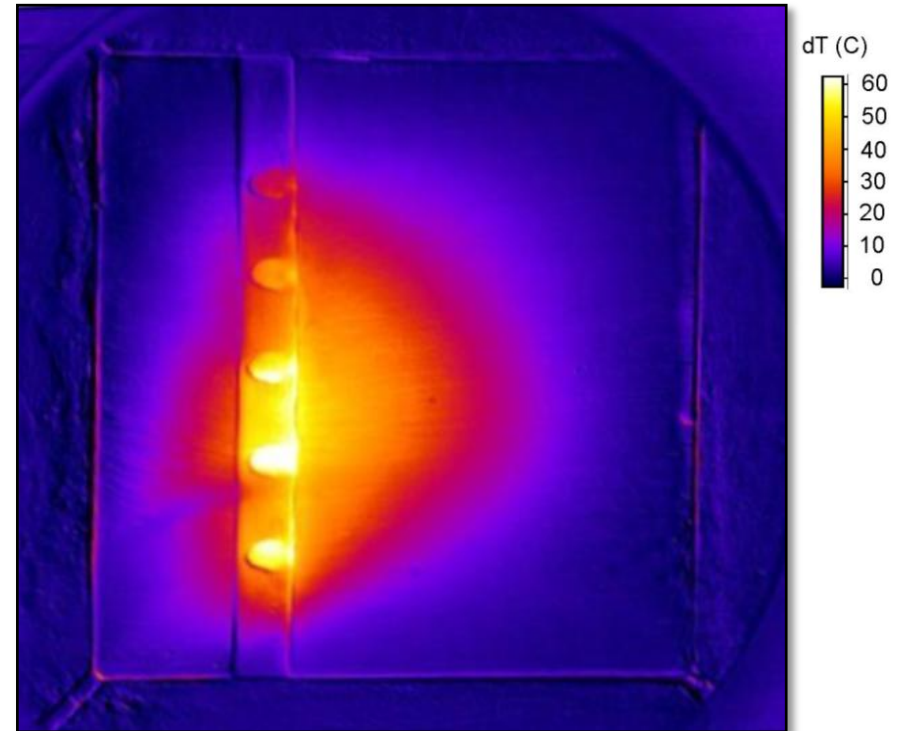


Preliminary CFD Results

Preliminary CFD Results



Experimental Results



Summary

- **NETL high temperature test facility is operational**
 - Some improvement required for non-contact IR temperature mapping
- **Preliminary film cooling tests completed**
 - University of Pittsburgh
- **In-situ LDV capability is in place**
 - Testing postponed due to laser repairs
- **Ash deposition capability is in place**
 - West Virginia University
- **Developing CFD capability using ANSYS/Fluent Software**
- **University and industrial collaboration is encouraged**